

Spacecraft Navigation using the VLBA

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Why do VLBI of Spacecraft?

Doppler and Ranging gives accurate distance to spacecraft

Not always enough for accurate navigation, especially linear orbit.

VLBI gives accurate position on plane of sky

Orthogonal to ranging, and critical in some cases

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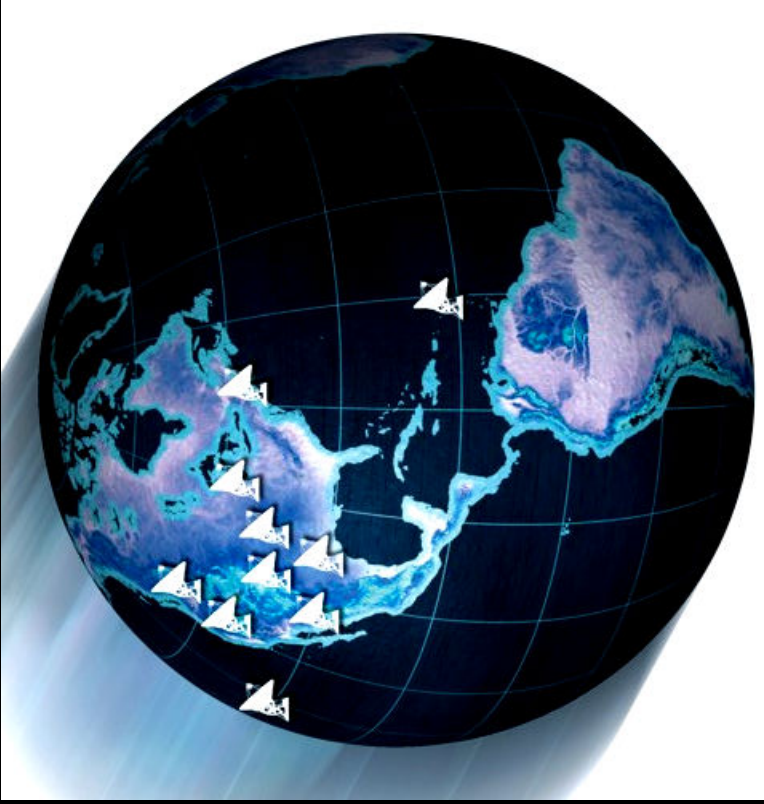
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Good news: Results even more accurate than anticipated.
Bad news: NASA has not yet decided to use VLBA for next
Mars mission (and many others) for navigation.

Phase Referencing to a Spacecraft



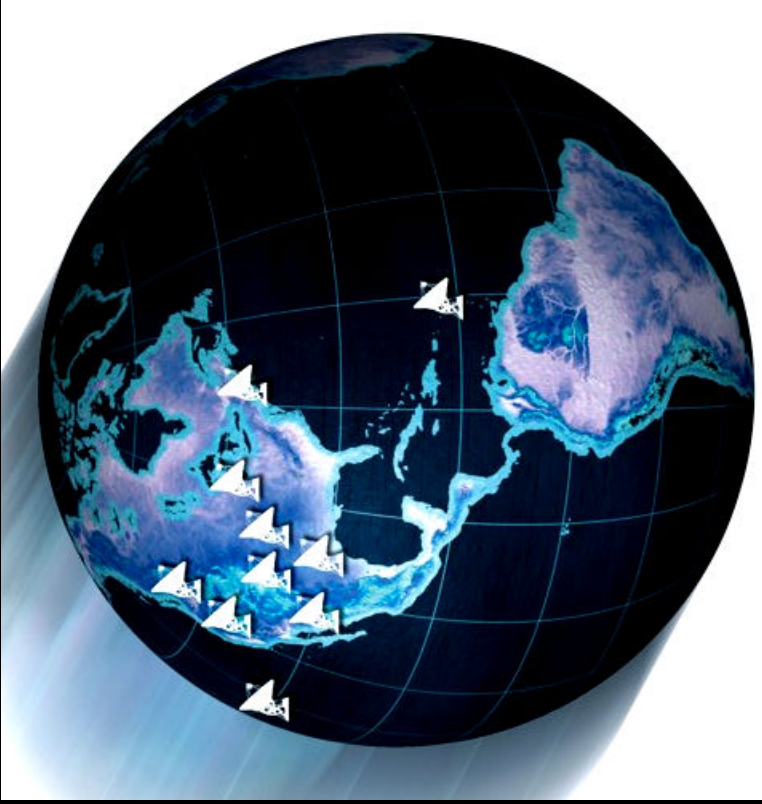
Alternate scans between calibrator and target

Use calibrator scans to determine clock
local troposphere etc, bandpass

Other special calibrations (measure
residual troposphere path delay)

Transfer calibrator phase to the target(s)

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Spacecraft 'normal' telemetry signal is no
different than a □ bright source with a
dominant somewhat variable spectral line.
>1 Jy, ~20 kHz wide, point source.

But, may have large, changing position
error (100 mas, 10 mas/hr in beginning)
wrt to input orbit.

Getting Spacecraft Positions

Make image of the target (spacecraft)
offset from phase center is spacecraft offset from input orbit,
assuming calibrator position is perfect.

Accuracy about ($<0.03 \text{ mas} * \text{cal-target separation in deg}$) 8 GHz

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JPL Navigators DO NOT BELIEVE image position offsets!

Depends on accuracy of correlator model which use:

- Orbit transfer from JPL to VLBA

- Curved wavefront

- Solar and planetary gravitational bending

- Parallax terms

Product to Navigators is measured TOTAL delay difference between calibrator
and target (same as for EVN, VLBA astrometric work)

Delay difference = Correlator model delay + calibrated delays
+ target phase / $2\pi v_0$

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Target phase must not have any 2π ambiguities! == good images)
Other arrays use group delays (phase slope with frequency)

Quasar



Phoenix Observing Strategy

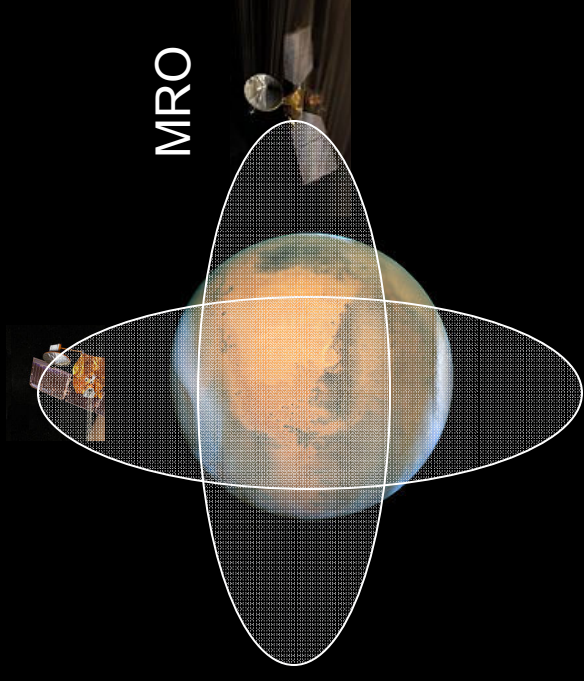


Phoenix



Phoenix Observing Strategy

Odyssey



MRO

Phoenix

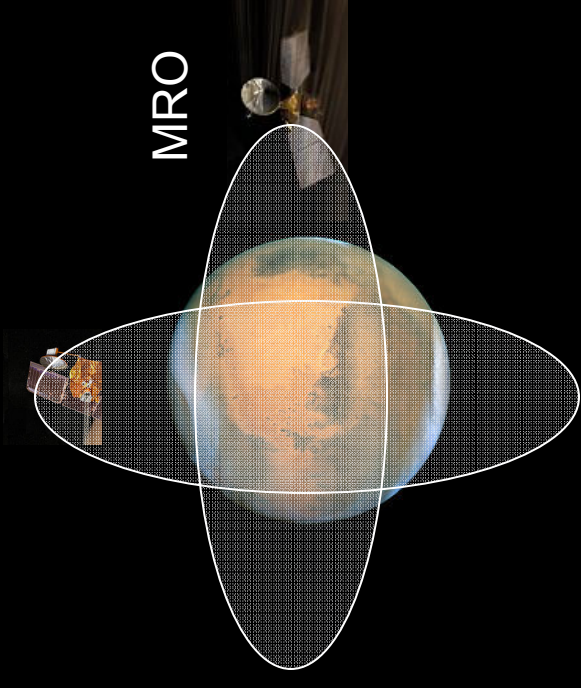


Quasar



Phoenix Observing Strategy

Odyssey



MRO

Phoenix



Observing Sequence:

- 1 min on quasar
- 1 min on Phoenix
- 1 min on Odyssey, MRO in-beam

Last two days before landing:

- Phoenix, Odyssey, MRO in-beam (<6')
- Three-hour sessions 8.4-8.6 GHz
- Ionosphere correction somewhat important



VLBA-Phoenix Position/Delay Analysis

Table of Phoenix position errors based on Navigator analysis of
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Date	Cal-Phoenix Separation (deg)	Orbiters-Phoenix Separation (deg)	Phoenix Acc from Cal Rel (mas)	Phoenix Acc from Orbiters Abs (mas)
Mar 01	2.3(1)	3.1	0.20 + 0.25	0.33
May 14	1.4(2)	1.8	0.09 + 0.15	0.20
May 17	1.5	1.3	0.13 + 0.15	0.14
May 19	1.6	0.8	0.08 + 0.15	0.07
May 22	1.8	0.22	0.10 + 0.15	0.05
May 23	1.8	0.14	0.06 + 0.15	0.03
May 25	1.9	0.06 *	0.09 + 0.15	0.02
May 26	1.9	0.02 *	0.11 + 0.15	0.015
(1) Calibrator is J0823+2223			* In beam	0.025 mas =
(2) Calibrator is J0842+1835				30 m at Mars

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Quasar-Phoenix accuracy limited by ICRF position uncertainty.
Phoenix-orbital accuracy limited only by angular separation.

Useful for any planet with orbiting spacecraft

Other Considerations

Occasionally need fast time response, <12 hours
e-VLBA (behind e-EVN), but MK, BR, HN critical stations

Upgrade VLBA to 33/8 GHz dual system:
NASA future telemetry frequency of 33 GHz.
8 GHz for Ionosphere correction
Move ICRF to 33 GHz (see poster)
Investigate phase stability, certainty of results

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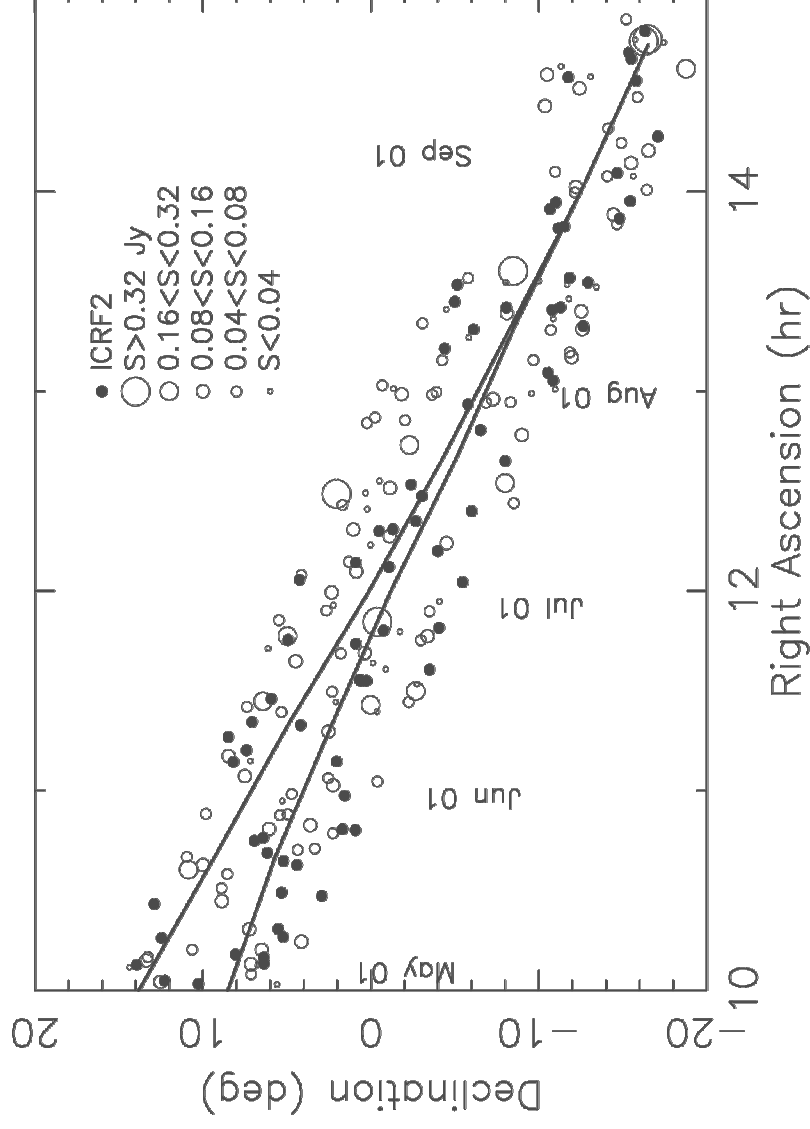
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Seamless operation that even JPL personnel could handle
Observation planning and strategy
Quick feed from telescopes into software correlator
Data reduction pipeline in AIPS,
special calibrations, images, delay determination

Transport delays to Navigation Package
Analysis of ranging and VLBI data.

Next Mars Mission MSL

Preparing to support Mars Science Laboratory Mission



Potential calibrators along path of MSL in 2011

Collaboration between NASA/NRAO not yet supported

2M US\$/year for ~25% of time for spacecraft/calibrator observations

Full e-VLBA capability

New 32-35 GHz receivers

Reason???? Such high accuracy not needed

Wind effect on Mars landing larger than 100 m

DSN 3-element array is good enough

Less impetus for high frequency ICRF

Still, other spacecraft programs

Determine Cassini position to establish Saturn ephemeris

Conclusions

Spacecraft Quasar accuracy similar to usual phase reference accuracy
better quasar position accuracy needed.

If planet has an existing radio emitter, much more accurate positions
of spacecraft wrt planet can be obtained.

Phase referencing can use 'normal' spacecraft telemetry signals.

Technique useful to use spacecraft for solar system dynamics.